



**Destination Green**

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# Progress on the development of the ICAO CO<sub>2</sub> Standard

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# Development of the CO<sub>2</sub> Standard



- The aircraft CO<sub>2</sub> Standard will result in a new Annex 16 Vol. III
- Two phases in the approach:
  - Phase 1 has been completed
    - Development of CO<sub>2</sub> Certification Requirement, including a CO<sub>2</sub> metric system and procedures.
  - Phase 2 is underway
    - CO<sub>2</sub> Standard setting process (stringency levels, technology responses, cost effectiveness assessments and interdependencies).



**The CAEP/9 meeting agreed on an Annex 16, Vol. III certification requirement.**



# This presentation...



- Principles of the CO<sub>2</sub> Standard development.
- Recent important decisions:
  - CO<sub>2</sub> metric system;
  - Certification requirement.
- Next steps and development timescales.



# CO<sub>2</sub> Standard Framework



## Aircraft CO<sub>2</sub> Standard

### **Certification requirement**

[Including: Metric system, procedures, measurement methodology, applicability]



### **Regulatory limit**

- Technology Standard similar to current Noise and LAQ Standards.
- Aircraft level Standard similar to Noise Standard.

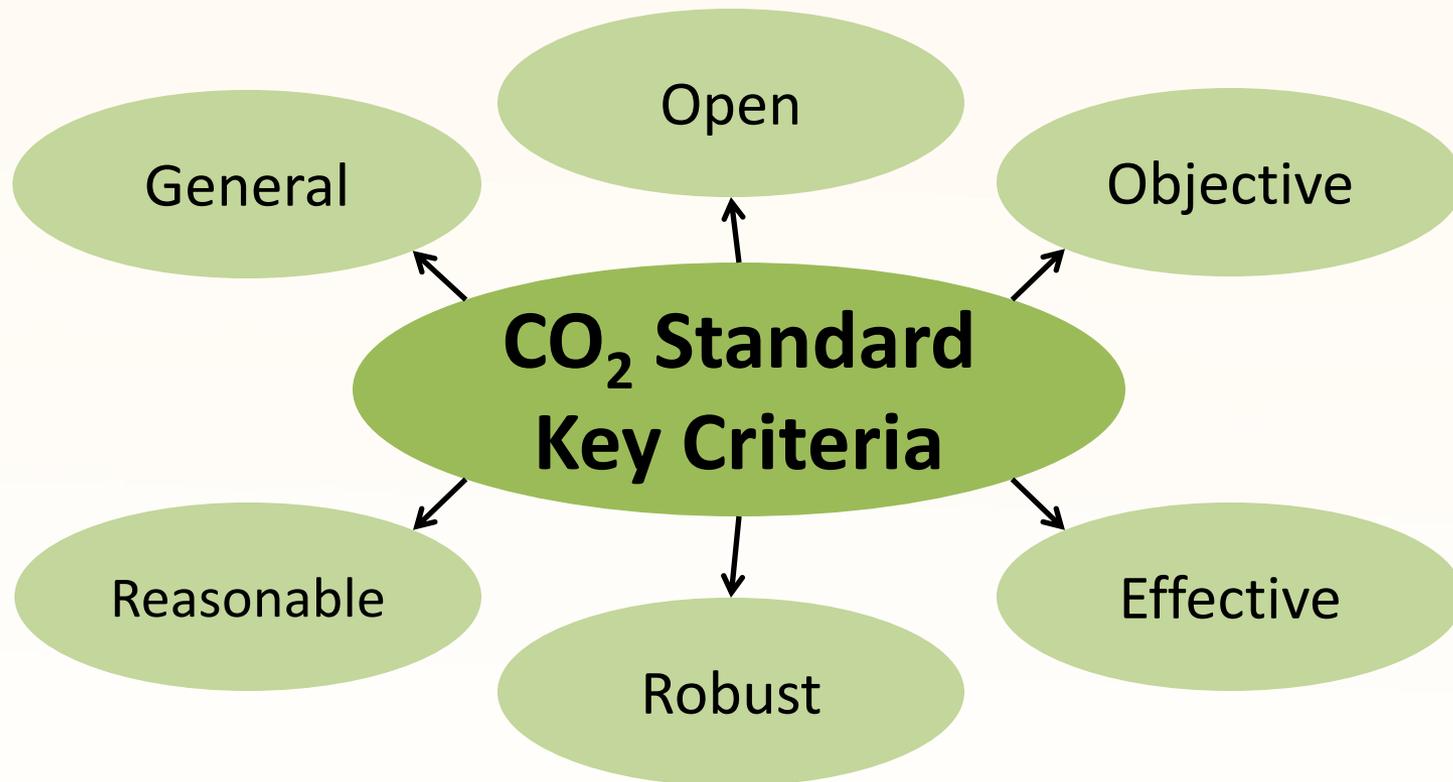


# High Level Principles



- An aircraft CO<sub>2</sub> Standard should focus on reducing CO<sub>2</sub> emissions through integration of fuel efficiency technologies into aeroplane type designs.
- To design a metric system which could permit transport capability neutrality at a system level when a stringency is applied based on this metric system.
- Aim for equitable recognition of fuel efficiency improvement technologies in an aircraft type design.

# Key Criteria





# Selection of a CO<sub>2</sub> metric system



- On 11 July 2012, at the CAEP Steering Group meeting in Saint Petersburg, an important step was made toward establishing the worldwide Aircraft CO<sub>2</sub> Emissions Standard.
- CAEP reached a unanimous agreement on a CO<sub>2</sub> metric system.
- The CO<sub>2</sub> metric system is a measure of aircraft fuel burn performance and therefore represents the CO<sub>2</sub> emissions produced by an aircraft.

**The CO<sub>2</sub> metric system will underpin the ICAO Aircraft CO<sub>2</sub> Emissions Standard**



# The principle of the CO<sub>2</sub> metric system



- The 'metric system' takes the form of:
  - A metric;
  - A correlating parameter; and
  - Certification test point(s).
- **The Metric is a function of:**
  - Cruise point fuel burn performance and aeroplane size.
- **The Correlating parameter:**
  - Maximum aeroplane mass.
- **The Certification test points:**
  - Three certification test points, each based on a fraction of aeroplane mass.



# Certification procedures



- Developed using a group of certification experts from States and international organisations.
- Resulting in the certification test and measurement criteria for determination and implementation of the CO<sub>2</sub> metric system.
- Development of procedures to measure the elements of the CO<sub>2</sub> metric system:
  - measurement of all parameters;
  - correction of measured data to reference conditions.



# Applicability and implementation



- The CO<sub>2</sub> Standard will be applicable at an aeroplane level.
- The CO<sub>2</sub> Standard will be applicable to new aeroplane types.
  - Discussions continue over including in-production types.
- The CO<sub>2</sub> Standard will be applicable to subsonic jet and propeller driven aeroplanes.
- Discussions over the applicability dates for the CO<sub>2</sub> Standard are continuing.



# CAEP approved Annex 16, Vol. III certification requirement



- The mature Annex 16, Vol. III certification requirement, which is based on the previously agreed CO<sub>2</sub> metric system, was approved by CAEP, including:
  - Part I - Definitions and symbols;
  - Part II - Certification standard for aeroplane CO<sub>2</sub> emissions based on the consumption of fuel;
  - Appendix 1 – Determination of aeroplane CO<sub>2</sub> emissions evaluation metric;
  - Appendix 2 – Calculating the parameter for aeroplane size.



# Taking stock of progress



- The CAEP agreed to take stock of the agreed CO<sub>2</sub> metric system and certification requirement before the final Standard was set.
- The CAEP agreed to publish the agreed CO<sub>2</sub> Standard certification requirement as an ICAO Circular for information-only, as soon as possible.



# Remaining work and timescale



- To finalise the CO<sub>2</sub> Standard the following issues remain:
  - definition of a no-change criteria;
  - applicability requirements;
  - regulatory limit;
  - applicability date for limit.
- The CAEP assessed a comprehensive work plan and agreed that the CO<sub>2</sub> Standard technical work would be finalised in late-2015.

# Thank you



For more information on the CO<sub>2</sub> Standard developments:

<http://www.icao.int/Newsroom/Pages/new-progress-on-aircraft-CO2-standard.aspx>



# BACKUP SLIDES ON THE CO<sub>2</sub> METRIC SYSTEM



# CAEP agreed CO<sub>2</sub> metric system



The Metric

The Correlating Parameter

**(1/SAR) / Б vs. MTOM**

**Cruise point fuel  
burn performance  
(1/SAR)**

Specific Air Range (SAR) is the distance an aircraft can fly for a given mass of fuel burned, measured at optimum conditions (speed, altitude, trim etc.).

**Aeroplane size  
(Б)**

Б is a function of Reference Geometry Factor (RGF).

RGF is a measure of fuselage size

**Aeroplane mass  
(MTOW)**

MTOM is the certified Maximum Take-Off Mass for the aeroplane type.

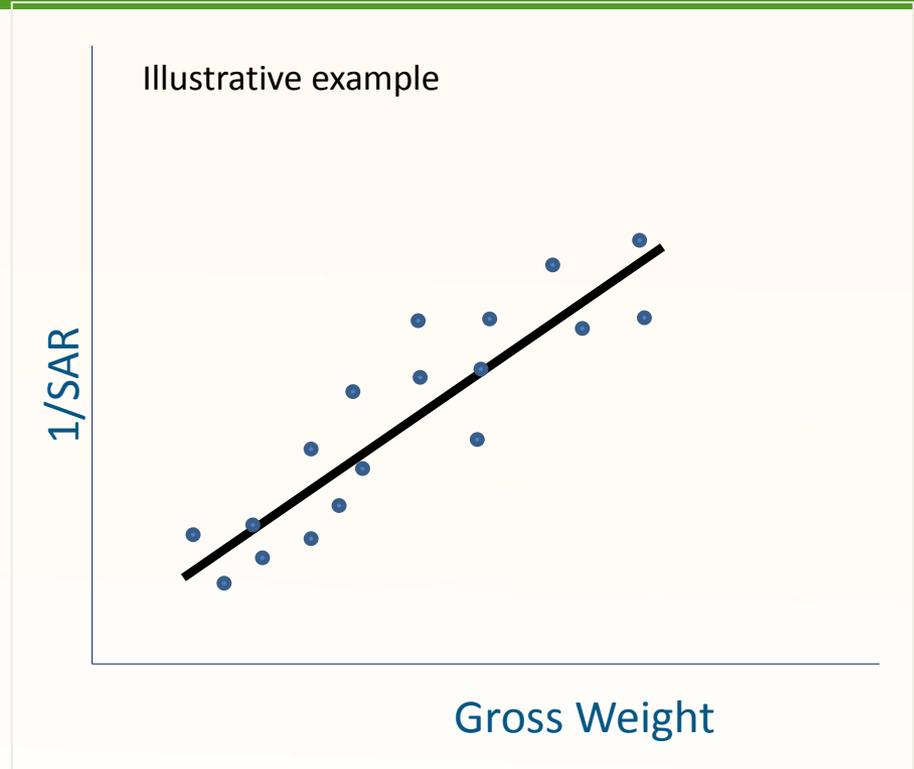
**Certification Test Points:** The CO<sub>2</sub> metric System will be evaluated at three aircraft gross masses, where each is represented by a fraction of MTOM.



# The Metric – Details on Specific Air Range (1/SAR)



- 1/SAR is a single point steady state instantaneous measure of fuel consumption per unit distance in cruise.
- **1/SAR = kg of fuel/km flown**
- Accounts for the main aeroplane design elements:
  - Propulsive efficiency
  - Aerodynamic efficiency
  - Aircraft weight changes



- 1/SAR varies primarily with aeroplane Gross Weight (GW).

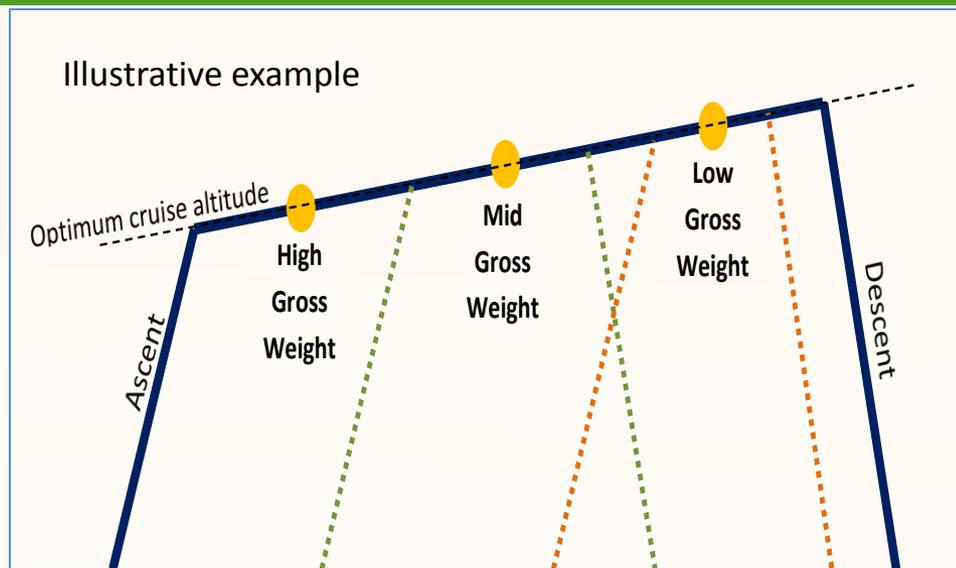


# The Metric – Details on Specific Air Range (1/SAR)



- 1/SAR is measured at three certification test points based on GW:

- High GW =  $0.92 * MTOM$
- Mid GW = Average of High GW and Low GW
- Low GW =  $(0.45 \times MTOM) + (0.63 \times (MTOM^{0.924}))$



- Each of these certification test points represents an aircraft cruise GW seen regularly in service.
  - The objective is to make the evaluation of fuel burn performance more relevant to day-to-day aircraft operations.
- The certification test points will be equally weighted to give a single metric value.



# The Metric – Details on Reference Geometric Factor (RGF)

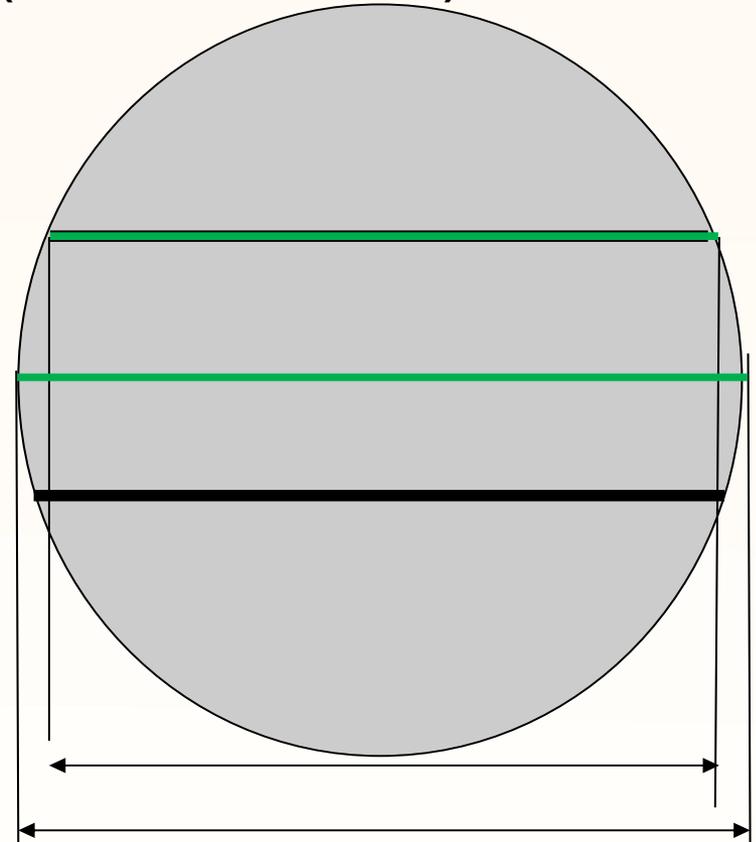
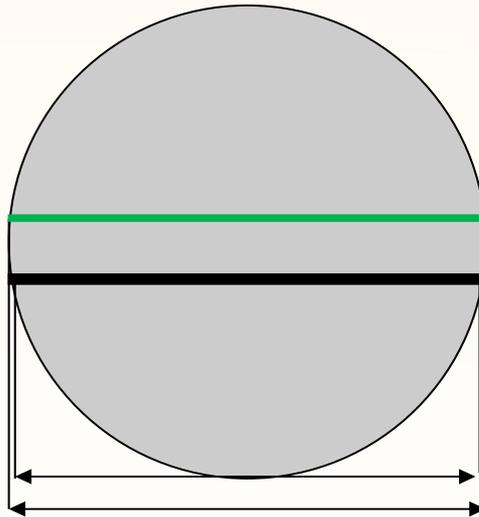
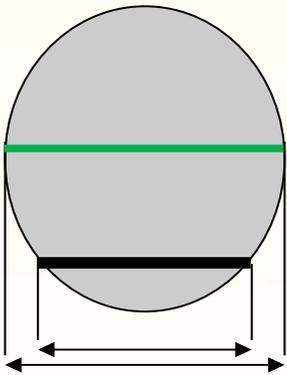


- The CO<sub>2</sub> metric system has been designed to be transport capability neutral, meaning:
  - In a transport capability neutral metric system aircraft types with diverse transport capabilities, but similar levels of fuel efficiency technology/design, have similar margins to the limit line.
- To achieve this (1/SAR) is augmented with RGF<sup>0.24</sup>.
- RGF is a measure of fuselage size.
- Accounts for instances when changes in aircraft size are not reflected by weight changes.
  - e.g. when an aircraft is a stretched version of an existing aircraft design.

# The Metric – Details on Reference Geometric Factor (RGF)

- **Boundaries of the Reference Geometric Factor (cross-sectional view)**

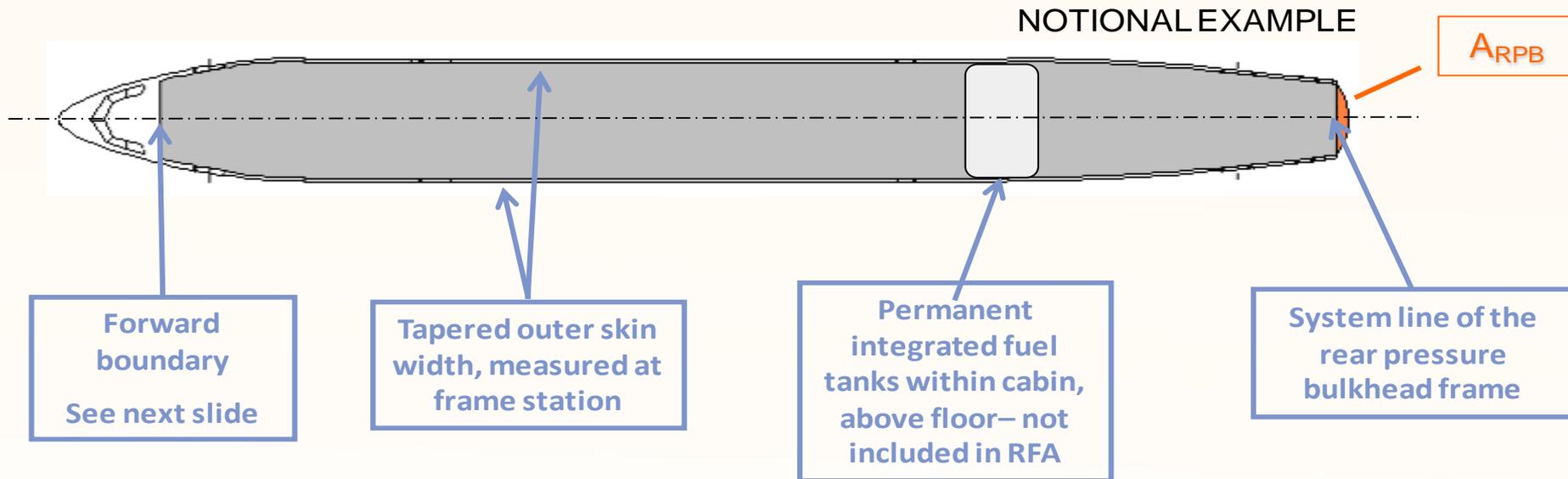
NOTIONAL EXAMPLES



 Reference Geometric Factor  
 Actual Floor Area

# The Metric – Details on Reference Geometric Factor (RGF)

## Boundaries of the Reference Geometric Factor (longitudinal view)



### Notes:

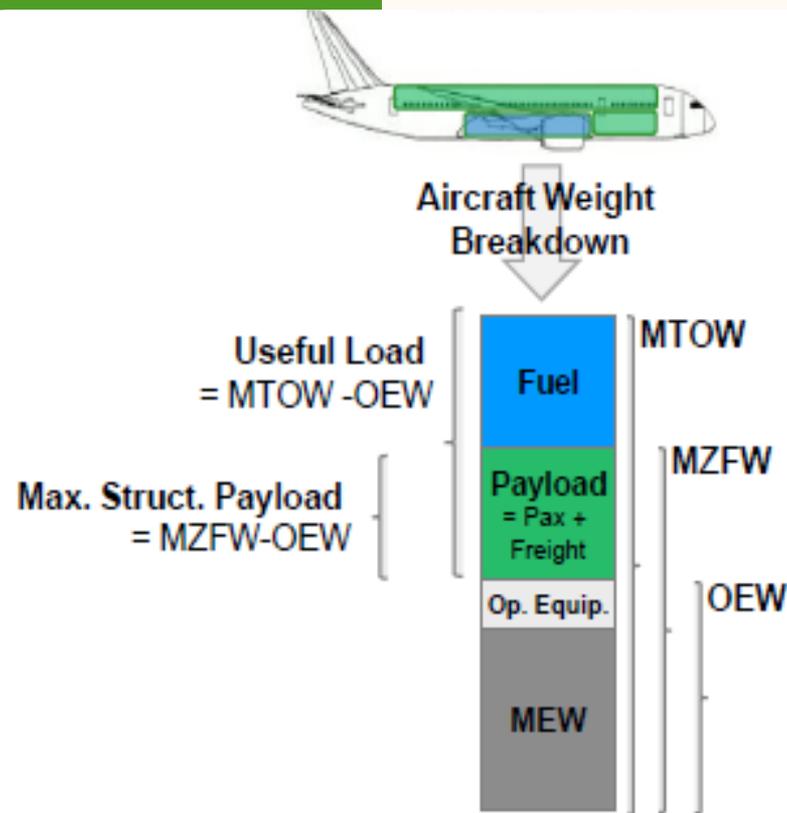
- Included
  - Aisles, assist space, passage ways, and stairwells
- Excluded
  - Rear Pressure Bulkhead dome
  - Crew rest or work areas which are not on the main deck or upper deck (ie 'loft' or underfloor crew areas)
  - Permanent integrated fuel tanks within cabin, above floor



# Correlating Parameter - Details on Maximum Take-Off Weight (MTOW)



- In many transport system metric systems, the correlating parameter is a measure of capability or capacity.
- This is not always the approach taken with ICAO environmental Standards.
  - The ICAO LAQ Standard uses engine Overall Pressure Ratio (OPR) to allow for other effects.
- In the CO<sub>2</sub> metric system, MTOW is used to represent the design of an aircraft type, accounting for the majority of features which allow an aircraft type to meet market demand.
  - MTOW is a certified value.



*MTOW: Maximum Take-Off Weight,  
MZFW: Maximum Zero Fuel Weight  
OEW: Operating Empty Weight,  
MEW: Manufacturer Empty Weight*



# CAEP agreed CO<sub>2</sub> metric system



## (1/SAR) / RGF<sup>0.24</sup> vs. MTOW

Evaluated at three certification test points:

$$\text{High GW} = 0.92 * \text{MTOW}$$

$$\text{Mid GW} = \text{Average of High GW and Low GW}$$

$$\text{Low GW} = (0.45 \times \text{MTOW}) + (0.63 \times (\text{MTOW}^{0.924}))$$

The three certification points will be equally weighted to derive a single metric system value.



# Summary of the CO<sub>2</sub> metric system



## (1/SAR) / RGF<sup>0.24</sup> vs. MTOW

Evaluated at three certification test points:

$$\text{High GW} = 0.92 * \text{MTOW}$$

$$\text{Mid GW} = \text{Average of High GW and Low GW}$$

$$\text{Low GW} = (0.45 \times \text{MTOW}) + (0.63 \times (\text{MTOW}^{0.924}))$$

The three certification points will be equally weighted to derive a single metric system value.

- The metric system aims to:
  - focus on reducing CO<sub>2</sub> emissions through integration of fuel efficiency technologies into aeroplane type designs;
  - equitably reward fuel efficiency improvement technologies in an aircraft type design.
  - be transport capability neutral;